

F.A.A. CRS OTUR438L 836 Texas Court, O'Fallon, MO 63366 636-978-5300 www.avmats.com

Pratt & Whitney Canada Engine JT15D-5A S/N: PCE-108335 TT: 7728.3 TC: 7504 TSO: 750.5 CSO: 533 Build Spec: 729

Repaired #2 Bearing failure and metal contamination. Disassembled, cleaned, inspected and reassembled engine. Replaced the following: 1 each Fuel Control Unit with overhauled P/N 3244851-10 S/N B29310; 1 each LPT Stator with overhauled P/N 3047462CL4070 FC:40.66; 1 each #4 Bearing with serviceable P/N 3101733-01 S/N FA9615721; 1 each AGB Input Driveshaft with overhauled P/N 3100641-01 S/N 5H840; 1 each AGB Duplex Bearing with overhauled P/N 3053963-01 S/N FC147645; 1 each #2 Bearing with overhauled P/N 3119579-01 S/N

The aircraft engine or appliance identified above was repaired and inspected in accordance with current regulations of the Federal Aviation Administration and is approved for return to service.

Pertinent details of the repair are on file at this repair station under work order No. 24535 Date 5/13/24

(Signature of Authorized Representative)

AVMATS Engine Support CRS OTUR 438L (Certificate No.)

836 Texas Court Lone Star Industrial Park O'Fallon, MO 63366

FAA0740010; 1 each Bleed Valve Control with serviceable P/N 800831-1-001; 1 each #1 Bearing with overhauled P/N 3115269-01 S/N FA9406710; 1 each #3 Bearing with overhauled P/N 3107493-01 S/N FC127545; 1 each #3½ Bearing with overhauled P/N 3101648-01 S/N FA7407D41; 1 each HPT Stator with overhauled P/N 3037261 S/N 7R417 FC: 39.19; 1 each Impeller with overhauled P/N 3040939 S/N A00020BX; 1 each HPT Disk with overhauled P/N 3036411 S/N A000098K; 1 each LPT 2<sup>nd</sup> Stage Disk with overhauled P/N 3022212 S/N 62A644; 1 each LPT 3<sup>rd</sup> Stage Disk with overhauled P/N 3029601. Performed satisfactory performance run and leak check at AES Test Facility. No discrepancies noted at this time. All work performed in accordance with Pratt & Whitney Canada JT15D-5A Overhaul Manual P/N 3037323, Revision 33 dated August 16, 2021 and Pratt & Whitney Canada JT15D-5A Maintenance Manual P/N 3037322, Revision 41 dated April 25, 2022. Note: S.B. 7363 incorporated.

JT15D ANALYSIS, PWC83021 Version 1.1.1 USER FILE: ABUS\_PRD\_85729\_WV\_8 FINAL ACCEPTANCE DATA ANALYSIS - REGULAR CALIBRATION FNGINE: 108335 BUILD: 1 1 1 CONDITION NUMBERS (10, 17, 3F) FULL CALIBRATION FINAL PERFORMANCE AT REFERENCE T.O. THRUST OF 2900 LBS PARAMETER VALUE AMALYSIS COD 15126.5 0.99375 BASED ON 5 POINTS B.F.3.L. BASED ON 4 POINTS 15115.952 0.99982 B.F.3.L. 4 REMOVED COMDITION NO. 30034.4 B.F.S.L. 1613.7 P.F.S.L. CURVE FIT SFC 8 2900 0.98392 DELO AVERAGE THEO 1.00928 AVERAGE HO\*RTHO/DO 81.855 P.F.S.L. 13.9853 W4\*RT4/P3 AVERAGE 1820.6 T5D/THO B.F.S.L. U FOINT TRIM 108.1 T7/TH0 1484.5 B.F.S.L. 0.64371 EFF (03) B.F.S.L. 9,5435 B.F.B.L. PS3/PTO 87.0 T7 SPREAD U FOINT NC REF = 91.58 4 M1 REF = 95.07 % NOTTLE CLASS . V4 PEF TS = 620.2 DEGREES C. SUMMARY PHC8:021 - 03-05-2024 / 08:15:53 BARO AV.TF ENGINE CELL MODET. 2024/05/03 14.46 1.08335 JT15D-5A SG TSG FNCOR .760 68. 1.034 LHV 10550. .760 PERFORMANCE DATA PEFERENCE THRUST - 2900. LES PAPAMETER ACTUAL PARAMETER ACTUAL 15116.0 SFC 8 2900. LBS .535 N1/PTHO 30034.4 SFC 8 N2/PTHO

1613.7 SFC 8

T5/TH0

## JT15D Life Limited Parts Record Sheet

Engine Model: <u>JT15D-5A</u> Build Spec: 729

TT: 7728.3

TC: 7504 TSO: 750.5

CSO: 533

11: //20.3	10.7501							
Nomenclature	Part Number	Serial Number	Total Time	Total Cycles	FCF per	Calcul a-ted	TC Remain-	Actual Cycles
Compressor Disks	Number	Tumber	2 137100	(Actual)	S.B.	Cycles	ing	Remain-
					7002			ing
Hub Low	3037911	42A942	7728.3	7504	1.0	7504	4496	4496
Compressor #1		***************************************		-		7.504	5405	7407
Hub Low boost compressor	3034112	41A945	7728.3	7504	1.0	7504	6496	6496
Impeller Centrifugal*	3040939	A00020BX	12684.4	9524	1.0	9524	2476	2476
Disk High Turbine	3036411	A000098K	9914.5	11108	1.0	11108	2892	2892
Rotor *			<b>.</b>			0006	5074	5074
Disk Low Turbine	3022212	62A644	8801.8	8026	1.0	8026	5974	5974
Rotor (Second) *								
Disk Low Turbine	3036113	62A966	8801.8	8026	1.0	8026	5974	5974
Rotor (Third) *					***			
High Turbine Blades	3029601						TT Rema	ıınıng:
(Reference S.B. 7003)*			***************************************			1 7 4		

Note: To determine calculated cycles multiply actual cycles times FCF (flight count factor).

To determine actual cycles remaining divide total cycles remaining by the flight count factor.

Remarks: As Installed

<sup>\*</sup> Indicates change at this repair.